

(19) World Intellectual Property
Organization
International Bureau



(43) International Publication Date
25 November 2004 (25.11.2004)

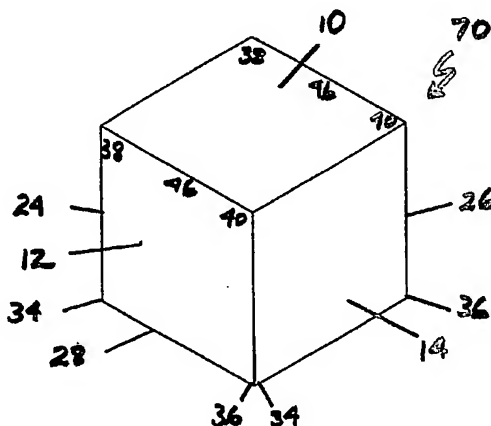
PCT

(10) International Publication Number
WO 2004/101216 A1

- (51) International Patent Classification⁷: **B23P 19/04**, (74) Agents: **CHRYSLIOU, Kerry et al.**; Chrysiliou Law, Level 2, 15-19 Parraween Street, Cremorne, Sydney, NSW 2090 (AU).
- (21) International Application Number: PCT/AU2004/000623 (81) Designated States (unless otherwise indicated, for every kind of national protection available): AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NA, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RU, SC, SD, SE, SG, SK, SL, SY, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, YU, ZA, ZM, ZW.
- (22) International Filing Date: 13 May 2004 (13.05.2004)
- (25) Filing Language: English
- (26) Publication Language: English
- (30) Priority Data: 2003902259 13 May 2003 (13.05.2003) AU (84) Designated States (unless otherwise indicated, for every kind of regional protection available): ARIPO (BW, GH, GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW), Eurasian (AM, AZ, BY, KG, KZ, MD, RU, TJ, TM), European (AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IT, LU, MC, NL, PL, PT, RO, SE, SI, SK, TR), OAPI (BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG).
- (71) Applicant (for all designated States except US): **TELEZYGOLOGY INC** [AU/AU]; Suite 1101, 61 Lavender Street, Milsons Point, NSW 2061 (AU).
- (72) Inventors; and
- (75) Inventors/Applicants (for US only): **RUDDUCK, Dickory** [AU/AU]; 81 Castle Circuit, Seaforth, NSW 2092 (AU). **WILSON, John, Ker** [AU/AU]; 52 Crestview Avenue, Kellyville, NSW 2155 (AU).
- Published:
— with international search report

[Continued on next page]

(54) Title: **IMPROVED ASSEMBLY AND DISASSEMBLY METHOD, SYSTEM AND COMPONENT**



(57) Abstract: The invention deals with methods for assembly and disassembly, especially suitable for assembly and disassembly by a computer. In relation to assembly, a component (10) is positioned in assembly (70) in accordance with a proposed position. The component (10) is confirmed as being in its correct spatial position in the assembly (70) by sensing means (38) or (46) capable of sensing spatial relationship and/or connection of the component (10) in relation to the assembly (70). Information regarding the sensed spatial relationship and/or connection is communicated to information processing means. A method for disassembly is also disclosed. The assembly and disassembly methods may require steps to be taken in a chosen sequence. The component (10) itself is also described, as is an improved assembly system which includes the component (10) together with information processing means.



For two-letter codes and other abbreviations, refer to the "Guidance Notes on Codes and Abbreviations" appearing at the beginning of each regular issue of the PCT Gazette.

Improved Assembly and Disassembly Method, System and Component

Technical Field

This invention relates to improved assembly and disassembly. The practical focus on the invention is on an improved system for use in conjunction with computer assembly and disassembly. However, the scope of the invention is not necessarily limited to this situation.

Background Art

Assembly is a process of joining components, often in an ordered sequence and dictated by a known set of steps. Likewise, disassembly often requires an ordered sequence.

- 10 Traditionally, the art of assembly or disassembly depends on a procedure or plan. In the simplest form, the assembly plan may be in the memory of a human assembler. Slightly more complex assemblies may require written instructions. More sophisticated assembly may require a detailed formal protocol, with each step in the assembly mapped out in detail, and using computer programs to control robotic installation.
- 15 Whether the assembly is simple, complex or sophisticated, and whether the assembled article is a simple toy, a piece of furniture or an automotive vehicle, for example, it is required that the components and sub-assemblies are in the correct spatial position and have the correct position relative to other components, sub-assemblies and to the assembly itself.

20 Disclosure of the Invention

It is an object of the present invention to facilitate assembly and/or disassembly and preferably to facilitate one or more of the following:

- (a) control of assembly and/or disassembly by computer program;
- (b) understanding of the relationship between the components;
- 25 (c) control of servicing by computer program;
- (d) control of inventory;
- (e) quality assurance; and
- (f) provision of alternate pathways to assembly and/or disassembly.

In a first aspect, this invention provides a method for assembling a component in an assembly, the method including the steps of:

- positioning the component in the assembly in accordance with a proposed position in the assembly;
- 5 sensing that the component is in its correct spatial position in the assembly by use of sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and
- communicating information regarding the sensed spatial relationship and/or connection to information processing means.

- 10 In a second aspect, this invention provides a method for disassembling a component from an assembly, the method including the steps of:

- removing the component from its position in the assembly;
- sensing that the component has been removed from its position in the assembly using sensing means capable of sensing spatial relationship and/or connection of
- 15 the component in relation to the assembly; and
- communicating information regarding the sensed spatial relationship and/or connection to information processing means.

In a third aspect, this invention provides a method for assembling a component in an assembly in a chosen sequence, the method including the steps of:

- 20 selecting or identifying the component;
- comparing the selected or identified component with information which correlates that component with sequence of assembly;
- positioning the component in the assembly in accordance with the sequence of assembly;
- 25 sensing that the component is in its correct spatial position in the assembly by use of sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and
- communicating information regarding the sensed spatial relationship and/or connection to information processing means.

In a fourth aspect, this invention provides a method for disassembling a component from an assembly in a chosen sequence, the method including the steps of:

- selecting or identifying the component;
- comparing the selected or identified component with information which
5 correlates that component with sequence of disassembly;
- removing the component from the assembly in accordance with the sequence of disassembly;
- sensing that the component has been removed from the assembly using sensing means capable of sensing spatial relationship and/or connection of the
10 component in relation to the assembly; and
- communicating information regarding the sensed spatial relationship and/or connection to information processing means.

Further, this invention provides a component for an assembly, the component being associated with:

- 15 a sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and
- means for communicating information regarding the sensed spatial relationship and/or connection of the component to information processing means.

This invention also provides an improved assembly system which includes:

- 20 the component of the invention; and
- information processing means for processing information communicated by the communication means.

The various aspects of the invention may be suitable for a wide range of uses, ranging from simple to complex to sophisticated systems. Preferably, however, the invention is
25 used for complex or sophisticated assemblies.

The assembly may be of any suitable type. Mentioned by way of example are building assemblies (including doors, windows, walls, etc), automotive vehicle assemblies and vehicle sub-assemblies, such as sound system installations, and aerospace assemblies. These are merely a small sample of the wide range of assemblies for which the
30 invention may be suitable.

The sensing means may be associated with every component in the assembly or with one or some only of those components. The sensing means may be chosen from any suitable sensing means. Some non-limiting examples are Hall Effect sensors, microswitches, optical fibre connections, optical sensors, capacitance detectors, radio frequency identification, proximity switches, electronic strain gauges and magnetic proximity detectors. By way of another example, the sensing means may involve a contact switch, i.e. connection of the component to an earthed circuit when the component is in place in the assembly. Movement of the component out of place (to a selected extent) can break the earth connection and signal lack of displacement, removal or improper positioning.

The sensing means may have more than one function. It may also form part of a fastener, to fasten the component to the assembly. The fastener may be a fastener disclosed by any of the following patent applications, the contents of which are imported herein by reference: International Patent Application No. PCT/AU99/00185 (Fixing and Release Systems), International Patent Application No. PCT/AU03/00759 (Further Improvements in Fixing and Release Systems), International Patent Application No. PCT/AU03/00933 (Improvements in Assembly and Disassembly), International Patent Application No. PCT/AU03/001539 (Bolt Assembly) and International Patent Application No. PCT/AU2004/000371 (Fastener for Air Bags and Other Uses).

There may be more than one sensing means per component. In this case, one sensing means may have the same or a different function from another. For example, a component having a square perimeter may have a first sensing means for sensing proximity of a corner of the component to the assembly. A second sensing means may sense proximity of an edge of the perimeter, not being located at a corner, to the assembly.

In addition to the spatial position/connection sensing means, the invention may include sensing means for other purposes, or the spatial position/connection sensing means may be capable of sensing other information in addition to spatial position/connection. For example, it may be desirable to sense the presence or absence of another component or element. It may be desirable to sense the identity of a component, so that it can be detected whether the component is a genuine component or not. It may be desirable to sense whether fasteners are locked or unlocked. These functions may be carried out by the sensing means in the method, system and component of the invention or by separate sensors.

The sensing means may sense the spatial relationship of the component to an element in the assembly or proposed assembly. The element may be another component in the

assembly or it may be something else, for example something related to the environment in which the assembly is to be inserted or used. As an example, if the assembly is to be placed in a frame or contained in a casing, the element may be all or a portion of that frame or casing.

- 5 The communication means may be any suitable means. Preferably, the communication means involves a computerised network.

The communications means may involve electrical, data and/or optical connections at relevant positions on the components of the assembly. Networking may involve hard wiring or wireless connection, using any suitable method.

- 10 The information processing means is preferably a computer which translates the information received via the communication means into a user-friendly form which is capable of display on the information display means (if present). The information display means is preferably a screen of suitable type, such as a liquid crystal display (LCD) screen, for example. The information display means is preferably included when
15 the invention is intended for use by humans.

Identification of a component may be desirable for some aspects of the invention. Identification means may include a serial number, a bar code, a radio frequency identification or any other suitable means.

- As indicated above, the invention has wide applicability. An example of how the
20 system of the invention can be employed is briefly described below in connection with the drawings, some of which are based on drawings in International Patent Application No. PCT/AU03/00759.

Brief Description of the Drawings

- The invention will be further illustrated by reference to the accompanying drawings.
25 The drawings show, inter alia, how the components in an assembly can relate to each other and how they can be assembled in two different forms to result in different types of assembly. The embodiments in the drawings are not limiting on the scope of the invention.

In the accompanying drawings:

- 30 Figure 1 shows in plan view six different components, which happen to be identical in this drawing;

Figure 2 shows in perspective view an assembly formed by the components in Figure 1, the assembly in this case being a cube;

Figure 3 shows the same six components of Figure 1 after forming into a different assembly, being a channel;

5 Figure 4 is a front elevation of an assembly being a vehicle dashboard facia;

Figure 5 is a side elevation of a fastener including sensing means;

Figure 6 is a cross-sectional view of the facia of Figure 4, taken along the lines 6-6 in Figure 4;

10 Figure 7 is a schematic diagram showing a plurality of the fasteners of Figure 5 connected by a multi-drop communications and power feed bus and controlled by a common control unit;

15 Figure 8 is a schematic diagram which is an alternate to that in Figure 7, showing a plurality of the fasteners of Figure 5 connected by a multi-drop RS-485 network and separate power feed bus and controlled by a computer via a RS-232 to RS-485 converter;

Figure 9 is a front perspective view of an assembly being an aircraft instrumentation panel;

Figure 10 shows the assembly of Figure 9 with an instrument released from the panel;

20 Figure 11 is a rear perspective view of the released assembly of Figure 10;

Figure 12 is a top plan view of a ring clip fastener in the locked state, suitable for use with the aircraft instrumentation panel of Figures 9 to 11;

Figure 13 is a side sectional view of the ring clip fastener of Figure 12, in the locked state;

25 Figure 14 is a top plan view of the fastener of Figures 12 and 13, but in the unlocked state;

Figure 15 is a cross-sectional view of the ring clip fastener of Figure 14 in the unlocked state and is useful for comparison with Figure 13;

Figure 16 is a top plan view of an alternate fastener which could be used in the assembly of Figure 9;

Figure 17 is a cross-sectional view of the fastener of Figure 16 in the locked state;

5 Figure 18 is the same cross-sectional view as that in Figure 17, but showing engagement of a peg;

Figure 19 is a cross-sectional view of the fastener of Figure 16 in the unlocked state and is useful for comparison for Figure 18; and

10 Figure 20 is a schematic diagram of an instrumentation servicing procedure suitable for disassembly of the assembly in Figure 9, using the fasteners of Figures 12 to 15 or of Figures 16 to 19.

Detailed Description of the Drawings

Referring first to Figure 1, there are shown six components 10, 12, 14, 16, 18 and 20. Each has a square perimeter with four sides 22, 24, 26 and 28 and four corners 30, 32,
15 34 and 36.

Each component 10, 12, 14, 16, 18 and 20 has sensing means 38, 40, 42 and 44, one located at each of corners 30, 32, 34 and 36, respectively. Each component 10 (etc.) also has edge sensors 46, 48, 50 and 52, located approximately in the centre of each of sides 22, 24, 26 and 28 respectively.

20 Located approximately in the centre of each component 10 (etc.) is a component identification means 54, 56, 58, 60, 62 and 64 respectively to identify each component.

Using a first computer program, components 10, 12, 14, 16, 18 and 20 are assembled to form an assembly, cube 70 shown in Figure 2. The computer (not shown) recognises the identity of each component 10 (etc.) through an identification sensor which sends an
25 enquiry to the identification means 54 (etc.). The computer compares the identity of the component 10 (etc.) with its data base which links identity with proposed position in the assembly 70. The components 10 (etc.) are positioned manually or robotically to the correct position in accordance with identity. The computer uses information relayed by the edge sensors 46, 48, 50 and 52 and corner sensor 38, 40, 42 and 44 to ensure that the
30 components 10 (etc.) are assembled in the desired configuration and with the desired tolerance of snug fit.

The identification means 54 (etc.) may be any suitable identification means. A non-limiting example is a serial number or a barcode. This may be a unique number or code which is programmed into the component as part of the manufacturing process.

It is to be appreciated that, although in the drawings the components are shown as being
5 identical, they may in fact have special characteristics in use. For instance, component 12 may be a wall with windows, component 14 may be a wall with a door and component 10 may be a ceiling or roof, so that cube 70, as shown in Figure 2, forms a house.

Components 10, 12, 14, 16, 18 and 20 may be used to form a different assembly, being
10 channel 72, shown in Figure 3. Once again, the computer, using a different program, identifies each individual component 10 (etc.) and employs the edge sensor 46 (etc.) and the corner sensors 38 (etc.) in order to construct the desired assembly 72.

Because of the information conveyed by the edge sensors 46 (etc.) and the corner sensors 38 (etc.), in the case of either the first or second program, the program can be
15 designed so that the components 10 (etc.) are assembled in a strict hierarchy or sequence. The relationship of the assembly to an outside element, such as a street when the assembly 70 in Figure 2 is a house, may also be determined. This can be easily understood if one refers to the example where the assembly 70 in Figure 2 is a house. During assembly, component 20 is first laid down to form a floor, then components 12,
20 14, 16 and 18 are erected to form vertical walls. Component 10 cannot be attached as the roof until components 12, 14, 16 and 18 are properly in place on floor 20. The assembly 70 is constructed so that component 14 is closest to the street (not shown) and so that component 14 is parallel to the street.

Referring now to Figures 4, 5 and 6, the vehicle dashboard 170 in Figures 4 and 6
25 contains a radio 172 held in fascia 170 by one or more overhang fasteners 160 (refer Figure 5). One such overhang fastener 160 is just discernible in Figure 6 and is quite small. Fastener 160 is not easily accessible from the outside of vehicle dashboard 170. Fastener 160 can be addressed via a microprocessor or connection to a circuit board (not shown) and can be activated so as to release radio 172 from its position in fascia 170.

30 Fastener 160 includes sensor 74 which senses whether fastener 160 is in the locked or unlocked position. As can be seen from Figure 6, dashboard assembly 170 includes component engagement sensor 76, for sensing whether the component, in this case radio 172, is in correct spatial position or is properly engaged in vehicle dashboard 170.

A plurality of fasteners 160 may be involved in retaining radio 172 in position in dashboard 170. The plurality of fasteners are individually addressable and may be connected by a multi-drop communications and power feed bus in the configuration shown conceptually in Figure 7. Any suitable number (n) of fasteners 160 may be
5 integrated in this configuration, which is controlled by a common control unit and power feed 80.

As a variation, as shown in Figure 8, fastener 160, again being individually addressable, may be connected by a multi-drop network 82 controlled by a computer 84, with a separate power supply 86.

- 10 A vehicle dashboard 170 such as that represented in Figures 4 and 6 may be designed so that the components can be assembled or disassembled only by use of a computer program. In order to remove and replace radio 172, intelligent fasteners 160 are instructed to release so that the radio 172 can be removed and replaced. On dashboard 170 may be a trim facing 78 which must be removed before radio 72 can be removed.
- 15 The trim facing 78 may be secured by more fasteners 160. In applying the method of the present invention, the computer program involved in the disassembly receives information as to whether trim facing 78 has been removed or not. For this purpose, magnetic proximity sensors (not shown) are arranged around the perimeter of trim facing 78, either integrated with fasteners 160 or as separate devices. The proximity
20 sensors convey information as to the proximity of each part of trim facing 78 to dashboard 170. The information is conveyed using the computer network and processed by the computer before being displayed on a screen for the computer operator.

When the new radio 172 is installed and the trim facing 78 is reassembled, the computer program can check for correct spatial position of all components, including trim facing
25 78. If the new radio 172 includes identification means and if an identity sensor is included, the program can also check to see if a genuine replacement component has been used and report on that aspect.

Reference is now made to Figures 9 to 11. Assembly 100 has a component being instrumentation panel 90 and a component being panel housing 92. Instrumentation
30 panel 90 includes various instruments (not shown) for the aircraft. Instrumentation panel 90 includes at least two fasteners 94 and a sensor 96 (refer Figure 11) for sensing whether or not instrumentation panel 90 is engaged with panel housing 92. Panel housing 92 includes fastener pegs 98 (refer Figure 10) for engagement in fasteners 94, as well as sensor 102, to sense whether instrumentation panel 90 is engaged with panel
35 housing 92. Sensors 96 and 102 may together form sensing means in accordance with the invention to sense connection between panel housing 92 and instrumentation panel

90. Sensing may be by completion of an electrical circuit, or by use of a reed switch or other suitable means.

Although Figures 10 and 11 show fastener peg 98 on panel housing 92 and fastener 94 as forming part of instrumentation panel 90, these components may be reversed.
5 However, it may be more convenient to include fasteners 94 on instrumentation panel 90 since they may be connected to the electrical circuitry in instrumentation panel 90.

As will be seen from Figure 11, panel housing 92 includes moulded panel receptacle 104, for snugly receiving instrumentation panel 90.

Fastener 94 is shown in more detail in Figures 12 to 15. The ring clip fastener 94
10 illustrated in these Figures is particularly suitable for securing components to a panel as in Figures 9 to 11. Fastener 94 is shown in the locked state in Figures 12 and 13 as engaging fastener peg 98 and in Figures 14 and 15 after release of fastener peg 98.

Fastener peg 98 includes groove 104. When peg 98 is pushed into fastener 94, arms 106 of bias spring 108 are pushed apart by ramp 110 on peg 98. Arms 106 snap fit into
15 groove 104 to lock peg 98 in fastener 94. Thus instrumentation panel 90 can be mounted in panel housing 92 by simple manual pressure.

Included in fastener 94 is engagement sensor 112 which senses contact with base 114 of peg 98. Engagement sensor 112 confirms engagement of peg 98 in fastener 94 and can communicate that information to a computer (not shown).

20 As can be seen from Figure 12, fastener 94 includes a shape memory alloy (SMA) wire 116. As has been described in the patent specifications imported herein by reference, a shape memory alloy wire can be configured so that it contracts when activated. In the case of fastener 94, the shape memory alloy wire 116 is connected to terminals 118 which bear on arms 106 of bias spring 108.

25 When appropriate energy is applied to SMA wire 116 - for example, electrical energy - and SMA wire 116 is heated to the necessary temperature, SMA wire 116 contracts as shown in Figure 14. Contraction of SMA wire 116 causes terminals 118 to move apart. Because terminals 118 bear on arms 106, those arms 106 are forced apart. Arms 106 move out of groove 104, permitting release of peg 98. Engagement sensor 112 can
30 report on the release of peg 98.

Included in fastener 94 are sensors 120 which sense the locked status of fastener 94. When fastener 94 is in the unlocked state as shown in Figure 14, it will be seen that

arms 106 contact sensors 120 and thus can complete a circuit or act as a reed switch so that sensors 120 can report on the locked or unlocked status of fastener 94.

Reference is now made to Figures 16 to 19. These illustrate a stud fastener which can be substituted for the ring clip fastener of Figures 12 to 15. The stud fastener of Figures 16 to 19 can be regarded as more robust and can carry a higher load compared to the ring fastener of the previous Figures.

Peg 98 of Figure 10 is substituted by peg 98a shown in Figures 18 and 19.

Fastener 130 of Figures 16 to 19 has a plurality of teeth 122 adapted to engage groove 104a in peg 98a. Teeth 122 are biased towards the locked position by tension spring 124. Tension spring 124 also served to keep teeth 122 in position. Peg 98a can be pushed into fastener 130. Ramp 110a on peg 98a will push against teeth 122 and against the tension of spring 124. As peg 98a is pushed into fastener 130, teeth 122 will spring back into groove 104a to lock peg 98a into fastener 130.

As in the fastener of Figures 12 to 15, fastener 130 includes engagement sensor 112. When contacted by base 114a of peg 98a, engagement sensor 112 can report connection of peg 98a with fastener 130 and hence connection of instrumentation panel 90 in panel housing 92.

Fastener 130 includes shape memory alloy wires 126. These are wound around shuttle 128 which is rotatable within body 132 of fastener 130. Shape memory alloy wires 126 are fastened at each end to a non-rotatable part of fastener 130 (not illustrated). When shape memory alloy wires 126 are caused to contract by the application of suitable energy to reach the necessary temperature, shuttle 128 rotates to the position shown in Figure 19. Shuttle 128 includes camming surface 134. As can be seen by Figure 19, camming surface 134 ramps down on tail 136 of tooth 122, pushing tail 136 against the bias provided by tension spring 124 and drawing each tooth 122 out of engagement with groove 104a. Thus peg 98a is freed from fastener 130. Tension spring 124 may also bias shape memory alloy wires 126 to their relaxed configuration.

Included in fastener 130 are lock status sensors 138 and 140. When these are separated as shown in Figures 17 and 18, the sensors report that fastener 130 is in the locked state. When the sensors make contact, as shown in Figure 19, the sensors report that fastener 130 is in the unlocked state. Sensors 138 and 140 may act as a reed switch, for example, or their contact may enable completion of an electrical circuit, to signal the unlocked state. Other configurations and means of sensing may of course be applicable.

Turning now to Figure 20, this illustrates two alternate servicing procedures, which can be used in relation to the disassembly illustrated in Figures 9 to 11, and using the fastener of Figures 12 to 15 or the fastener of Figures 16 to 19. The servicing procedure described in connection with Figure 20 contains many steps, such as password entry,
5 etc., which are not essential to the invention but which may be highly desirable in an environment which requires high degrees of security and safety, such as the aircraft industry.

In Figure 20, the servicing is in respect of a multiple display unit (MDU), such as instrumentation panel 90 in Figure 9. Service is required as the MDU has some type of
10 malfunction and needs to be serviced or replaced.

To commence the procedure, a service person, using a computer, enters the service person's user name, password and location in the computer. The computer may form part of the aircraft or may be a laptop plugged into an appropriate port in the aircraft, or may be a PDA computer communicating with the assembly by, for example, infrared or
15 Blue tooth means.

The service person then enters such details as (in this case) that the maintenance is unscheduled, the aircraft registration number and a work order identification number.

Now, the service procedure follows the lefthand path or the righthand path in Figure 20, depending on whether the MDU is to be serviced or replaced.

20 If the MDU is to be replaced, the righthand path is followed. The service person enters into the computer the option that the MDU is to be replaced and also enters the new part number of the replacement MDU. At this stage, the program may be set up to refuse to continue if the new part number does not indicate that the replacement MDU is a genuine part.

25 Provided the new part number is acceptable, the computer instructs the fasteners to unlock to release the MDU to be replaced. In the case of the fasteners in Figures 12 to 15 or Figures 16 to 19, the computer activates the shape memory alloy wires to contract, to enable release.

Next, the computer instructs the service person to remove the old MDU assembly. If the
30 service person enters "No" or does not enter "Yes", the computer continues to instruct the service person to remove the old MDU assembly. If the service person enters "Yes", the computer checks that the old MDU assembly has in fact been removed, for example by receiving data from engagement sensor 112 in Figure 18.

Next, the computer instructs the service person to install the new MDU assembly and asks that this is confirmed. Provided the service person enters "Yes", the computer checks that the MDU has in fact been assembled to panel housing 92, once again by receiving information from a sensor, such as sensor 112. Optionally, if the new MDU
5 has embedded identification which can be read by the computer, the computer will compare the new part number entered at the beginning of the procedure with the actual part number on the new MDU and may reject the new MDU if there is not conformity (this option is not illustrated in Figure 20).

The computer then records in a log that the MDU has been replaced together with the
10 identification of the new MDU and the assembly is complete.

For servicing of an MDU without replacement, the lefthand path in Figure 20 is followed. The service person selects the option "disassemble MDU" and the MDU assembly is released by the fasteners unlocking, as already described. The service person is instructed to remove the MDU assembly and to confirm, as before.

15 The MDU itself contains a printed circuitboard (PCB) which is retained behind an access panel (not illustrated in Figures 9 to 11). The procedure in relation to the removal of the access panels from the MDU is preferably carried out using a PDA which does not require hard wiring. However, the procedure can be adapted to hard wiring if required.

20 The service person instructs the computer to release the access panel at the rear of the MDU and the fasteners, which may be the same type of fasteners as those already illustrated, release this panel. The service person is instructed to remove the access panel and to confirm. The service person then instructs the computer to release the PCB. The computer can instruct fasteners, of the same type as described earlier, to release in
25 order to allow the service person to remove the PCB.

At this stage, disassembly is complete and a log entry may be made. The PCB may be replaced and the procedure generally reversed to reassemble the MDU.

Industrial Applicability

The invention has wide industrial applicability, as will be readily apparent to one skilled
30 in the art. As only one aspect of that industrial applicability, the invention enables tasks which are currently carried out by highly skilled personnel to be conducted safely and securely by less skilled personnel. This is readily appreciated by the instrumentation servicing procedure exemplified in Figure 20.

CLAIMS

1. A method for assembling a component in an assembly, the method including the steps of:
 - 5 positioning the component in the assembly in accordance with a proposed position in the assembly;
 - sensing that the component is in its correct spatial position in the assembly by use of sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and
 - 10 communicating information regarding the sensed spatial relationship and/or connection to information processing means.
2. A method for disassembling a component from an assembly, the method including the steps of:
 - removing the component from its position in the assembly;
 - 15 sensing that the component has been removed from its position in the assembly using sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and
 - communicating information regarding the sensed spatial relationship and/or connection to information processing means.
- 20 3. A method for assembling a component in an assembly in a chosen sequence, the method including the steps of:
 - selecting or identifying the component;
 - comparing the selected or identified component with information which correlates that component with sequence of assembly;
 - 25 positioning the component in the assembly in accordance with the sequence of assembly;
 - sensing that the component is in its correct spatial position in the assembly by use of sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and

communicating information regarding the sensed spatial relationship and/or connection to information processing means.

4. A method for disassembling a component from an assembly in a chosen sequence, the method including the steps of:

5 selecting or identifying the component;

 comparing the selected or identified component with information which correlates that component with sequence of disassembly;

 removing the component from the assembly in accordance with the sequence of disassembly;

10 sensing that the component has been removed from the assembly using sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and

 communicating information regarding the sensed spatial relationship and/or connection to information processing means.

15 5. The method of claim 3, which includes the further step of halting the sequence of assembly unless the information communicated to the information processing means verifies that the component is in its correct spatial position.

6. The method of claim 4, which includes the further step of halting the sequence of disassembly unless the information communicated to the information processing
20 means verifies that the component has been removed from the assembly.

7. The method of any one of claims 1 to 6 which includes the further step of displaying information communicated to the information processing means on information display means.

8. The method of any one of claims 1 to 7 wherein the information processing means
25 is a computer.

9. A component for an assembly, the component being associated with:

 a sensing means capable of sensing spatial relationship and/or connection of the component in relation to the assembly; and

30 means for communicating information regarding the sensed spatial relationship and/or connection of the component to information processing means.

10. The component of claim 9 wherein the sensing means includes a Hall Effect sensor, a microswitch, an optical fibre connection, an optical sensor, a capacitance detector, a radio frequency identification, a proximity switch, an electronic strain gauge or a magnetic proximity detector.
- 5 11. The component of claim 9, wherein the sensing means includes a contact switch.
12. The component of any one of claims 9 to 11 wherein the sensing means is capable of sensing other information in addition to spatial relationship and/or connection.
13. The component of any one of claims 9 to 11 which includes a second sensor capable of sensing other information not being spatial relationship and/or
10 connection.
14. The component of claim 12 or 13 wherein the other information is:
 - (i) identity of the component or the assembly or part thereof; or
 - (ii) locked status of the component to the assembly.
- 15 15. The component of claim any one of claims 9 to 14, which has more than one of the sensing means.
16. The component of claim 15, wherein one of the sensing means is capable of sensing a different spatial relationship or connection from another of the sensing means.
17. The component of any one of claims 9 to 16, wherein the sensing means forms part of a fastener.
- 20 18. The component of claim 13, wherein the other information is the identity, presence or absence of another component.
19. An improved assembly system which includes:
 - the component as claimed in any one of claims 9 to 18; and
 - information processing means for processing information communicated by the
25 communication means.
20. The system of claim 19 which further includes:
 - information display means for displaying information processed by the
information processing means.

21. A method for assembling a component in an assembly substantially as herein described with reference to Figures 1 to 3 or 4 to 8 or 9 to 11 or 20 of the accompanying drawings.
22. A method for disassembling a component from an assembly substantially as herein
5 described with reference to Figures 1 to 3 or 4 to 8 or 9 to 11 or 20 of the accompanying drawings.
23. A component for an assembly substantially as herein described with reference to Figures 1 to 3 or 4 to 8 or 9 to 11 or 12 to 15 or 16 to 19 of the accompanying drawings.
- 10 24. An improved assembly system substantially as herein described with reference to Figures 1 to 3 or 4 to 8 or 9 to 11 or 12 to 15 or 16 to 19 of the accompanying drawings.

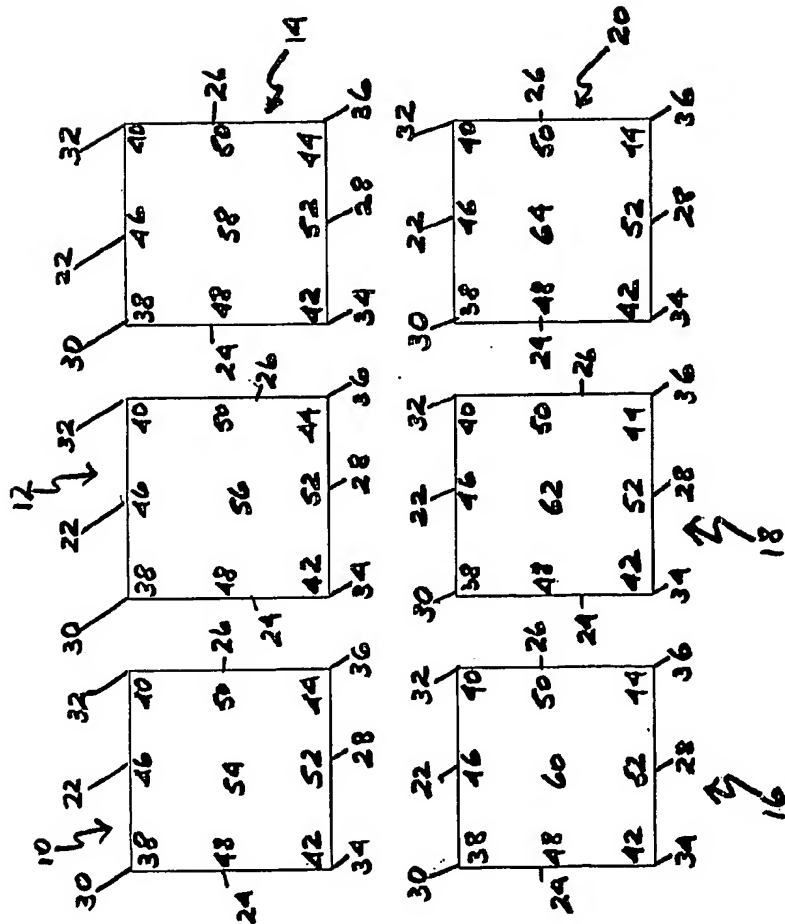


FIGURE 1

FIGURE 2

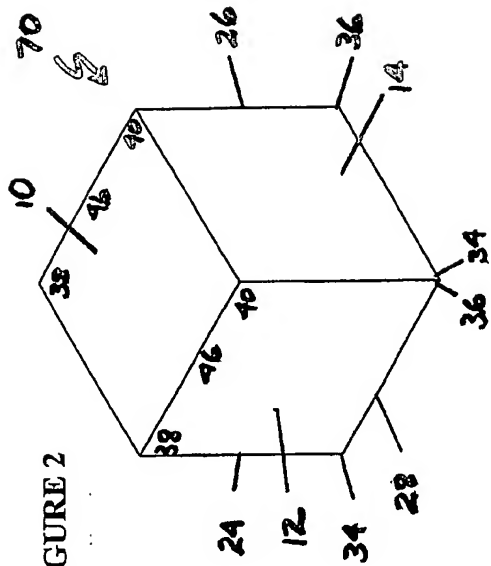
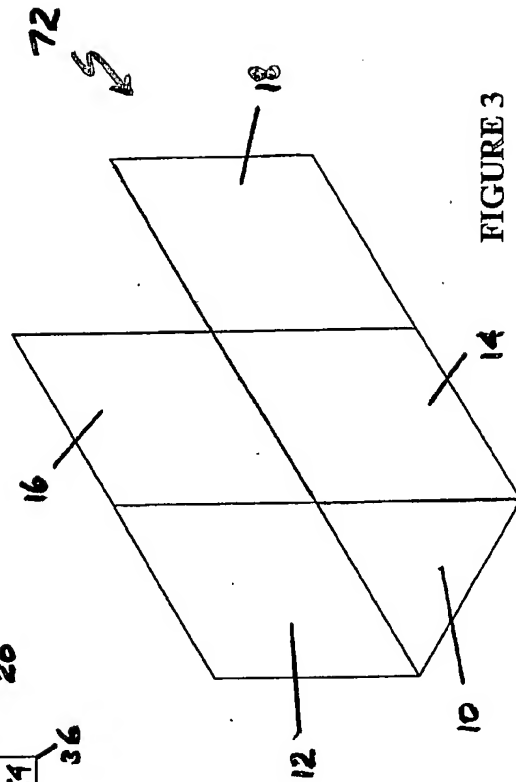


FIGURE 3



2/11

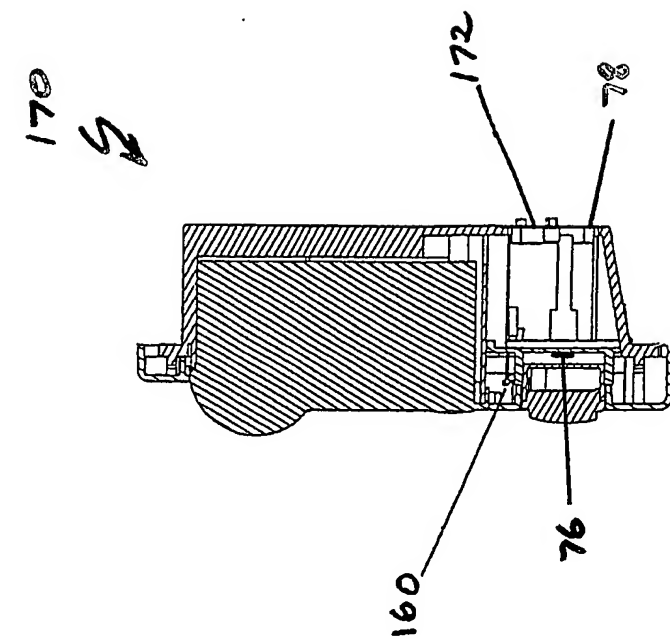


FIGURE 5

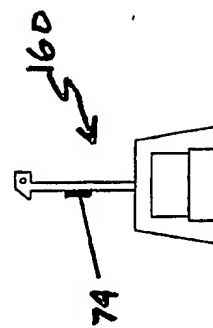


FIGURE 6

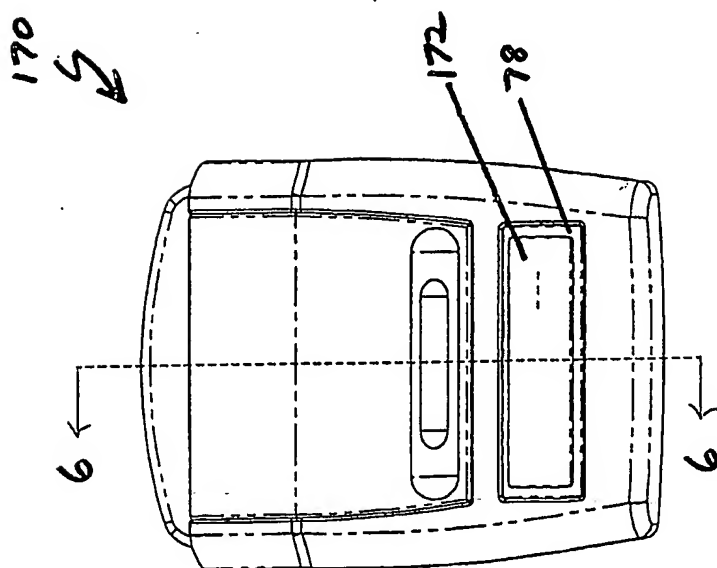


FIGURE 4

3/11

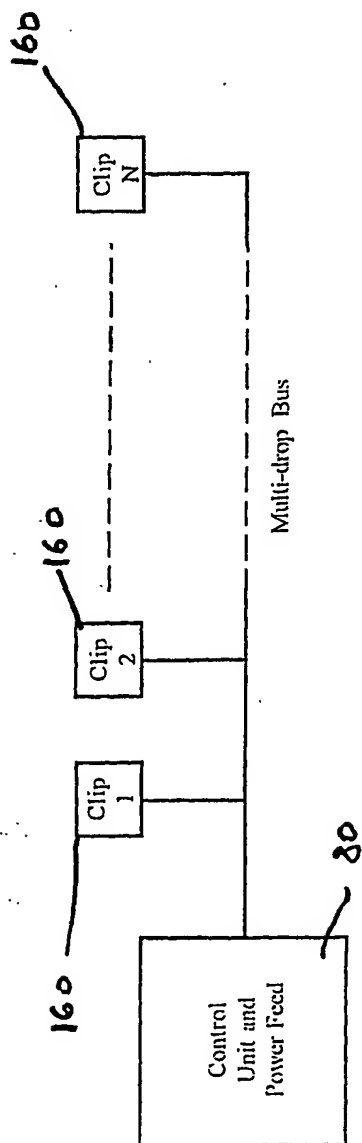


FIGURE 7

4/11

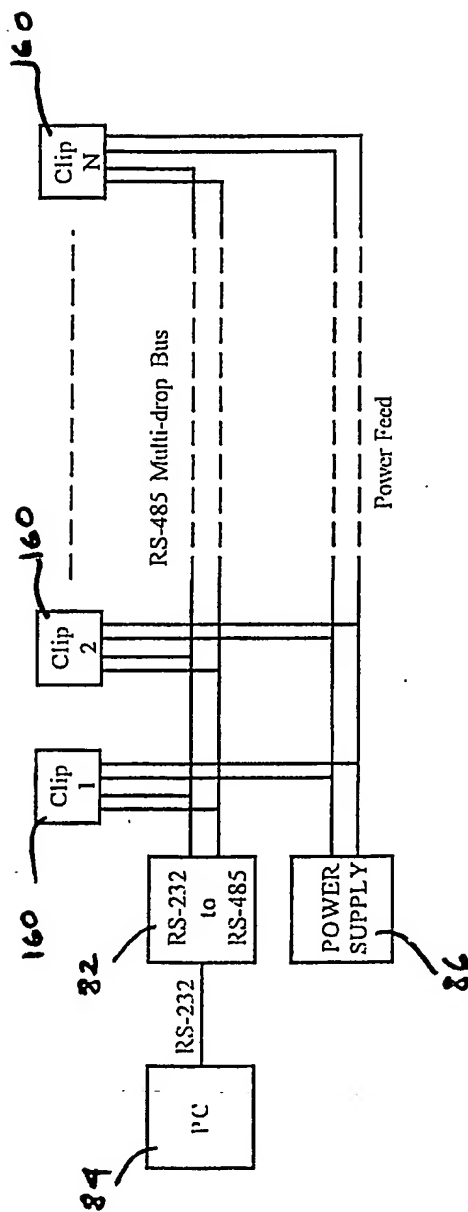


FIGURE 8

5/11

FIGURE 10

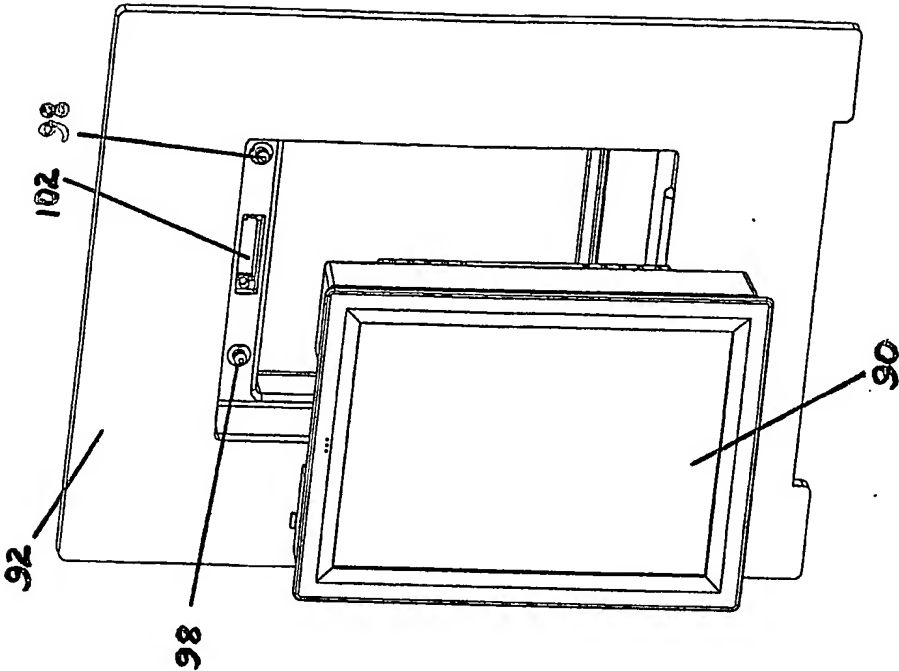
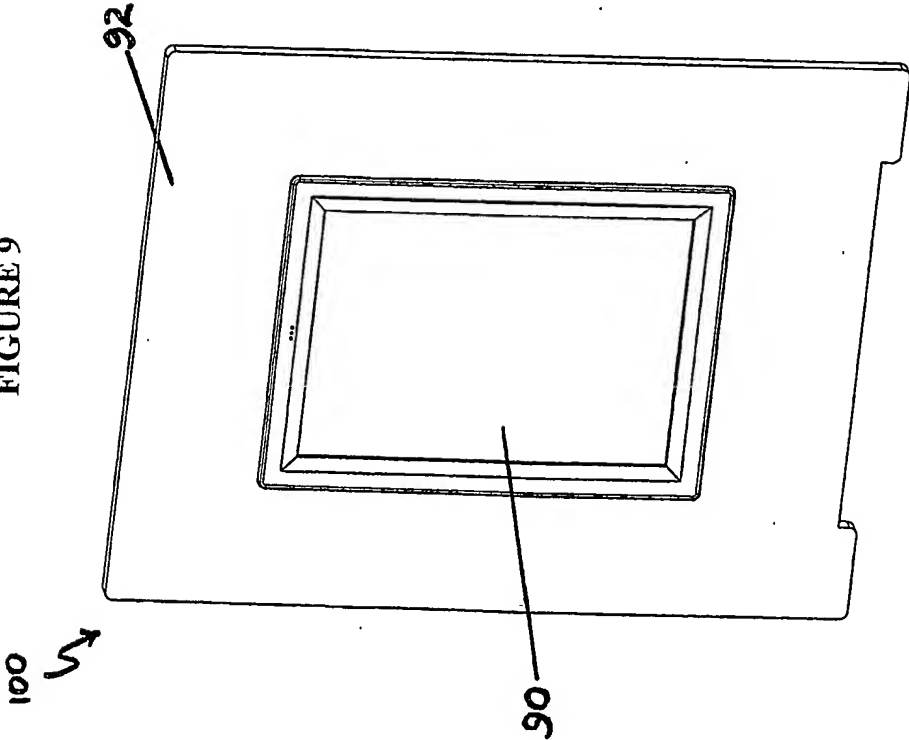


FIGURE 9



6/11

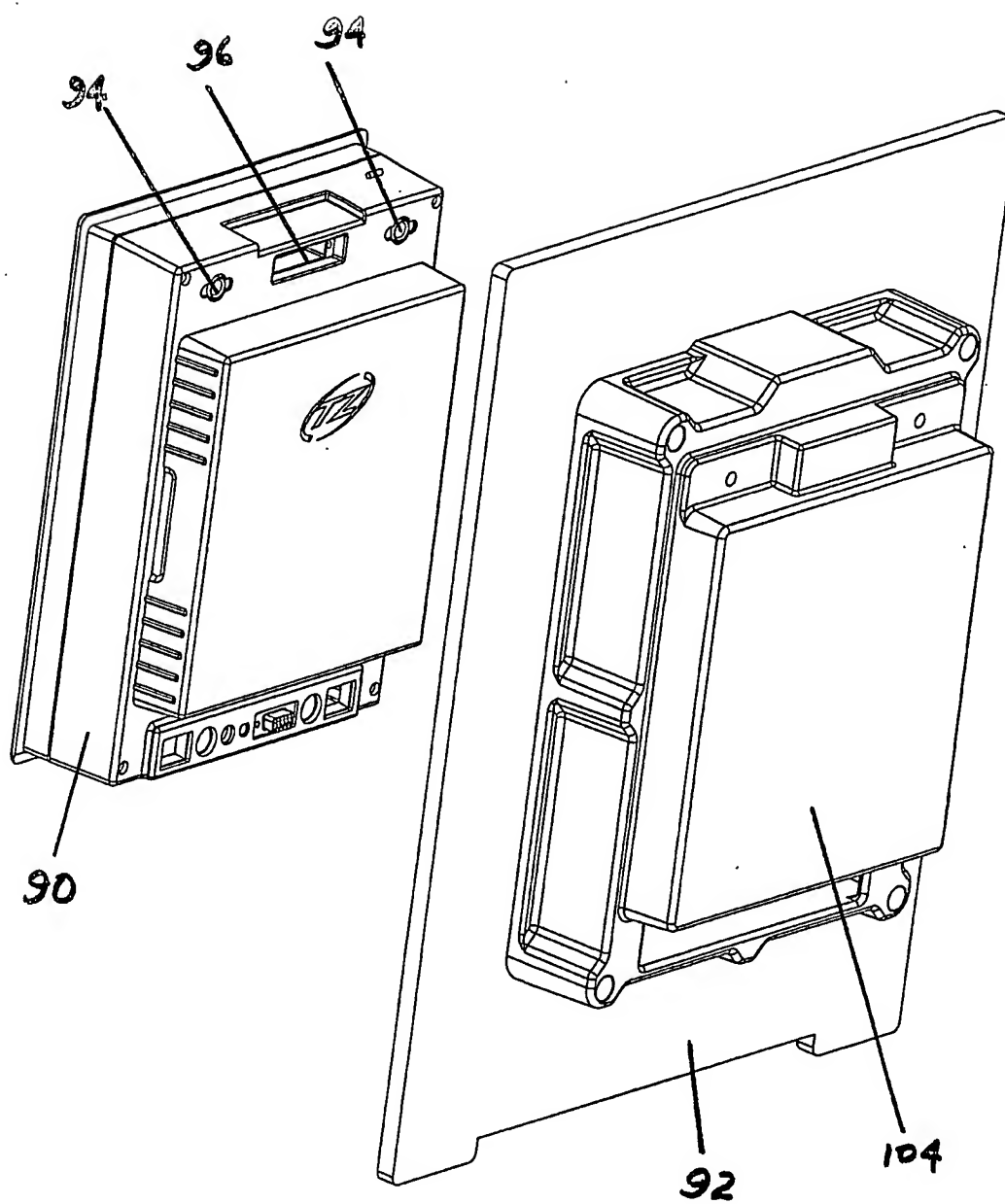


FIGURE 11

7/11

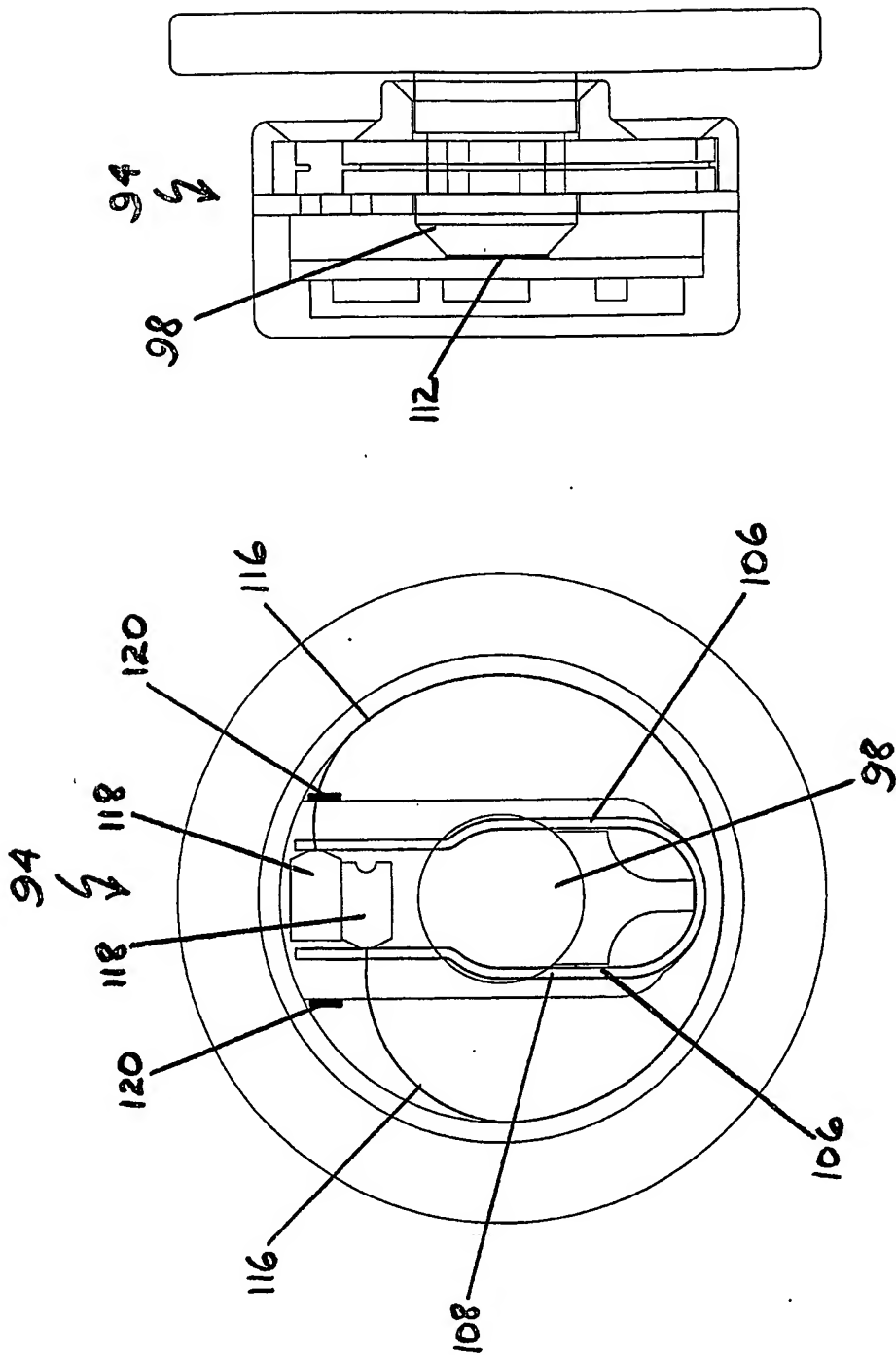


FIGURE 13

FIGURE 12

8/11

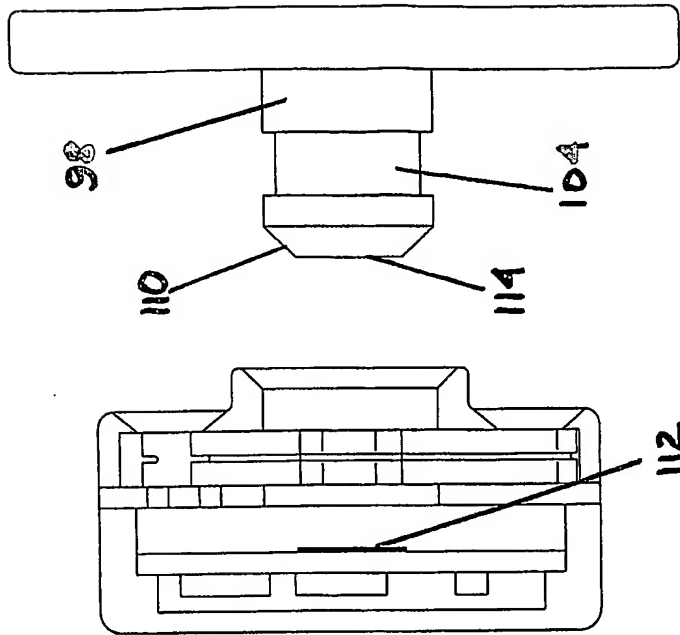


FIGURE 15.

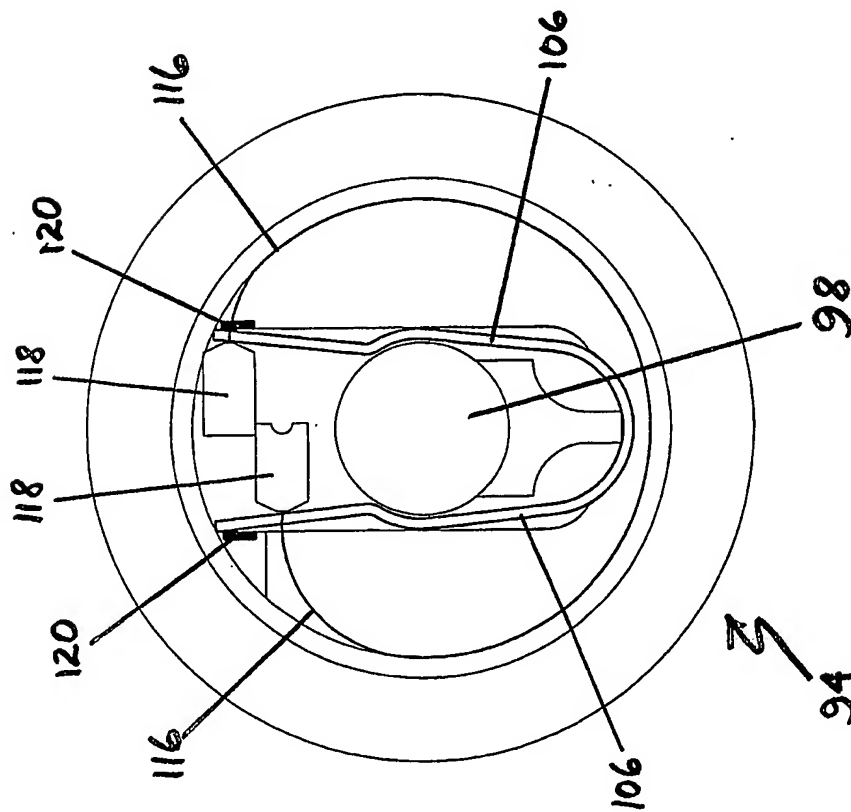


FIGURE 14

9/11

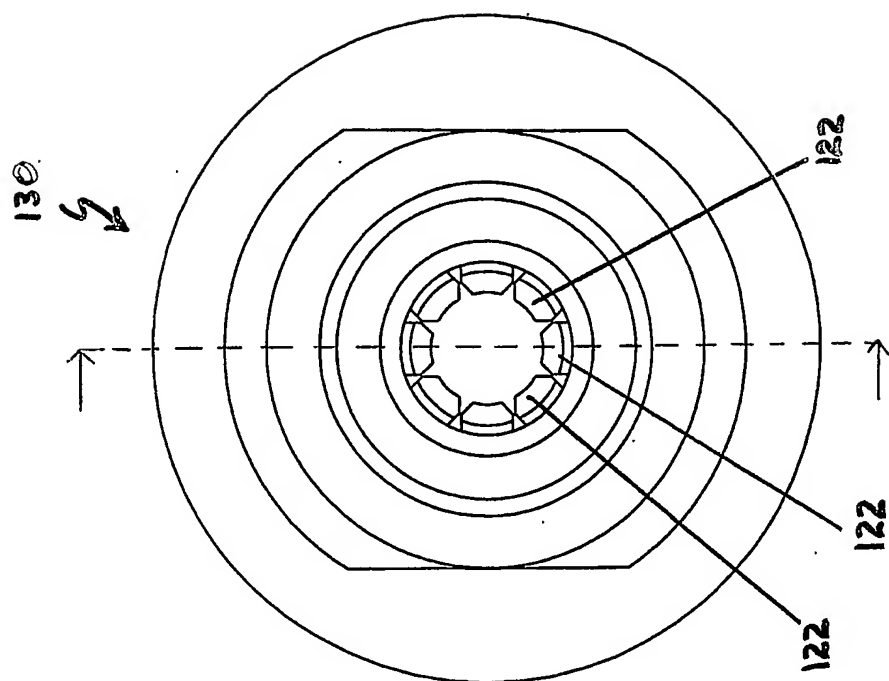


FIGURE 16

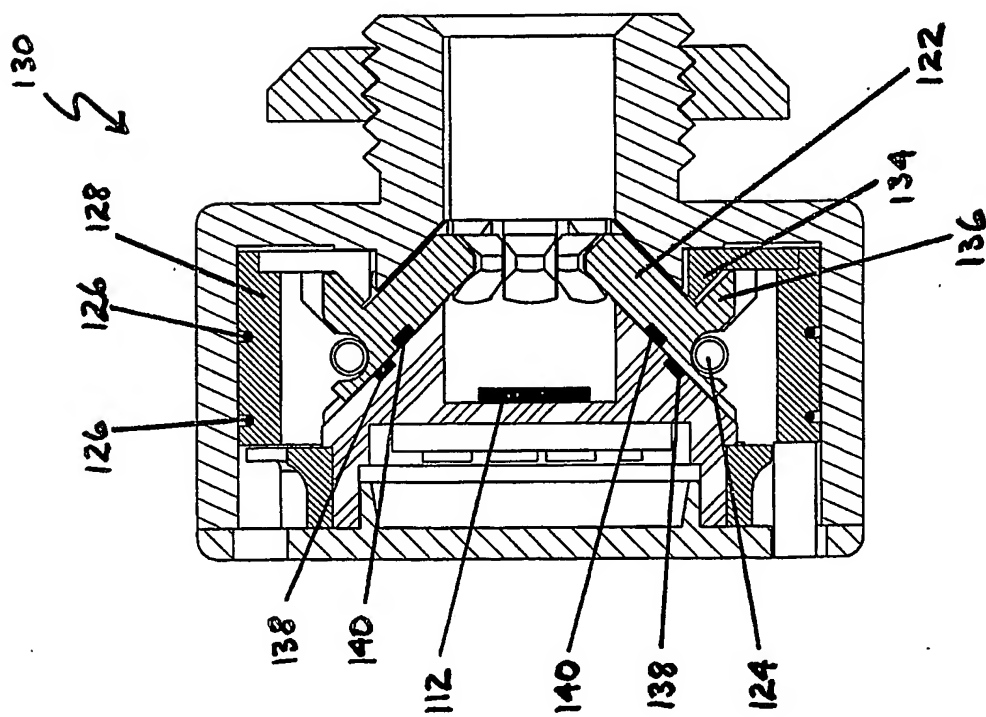


FIGURE 17

10/11

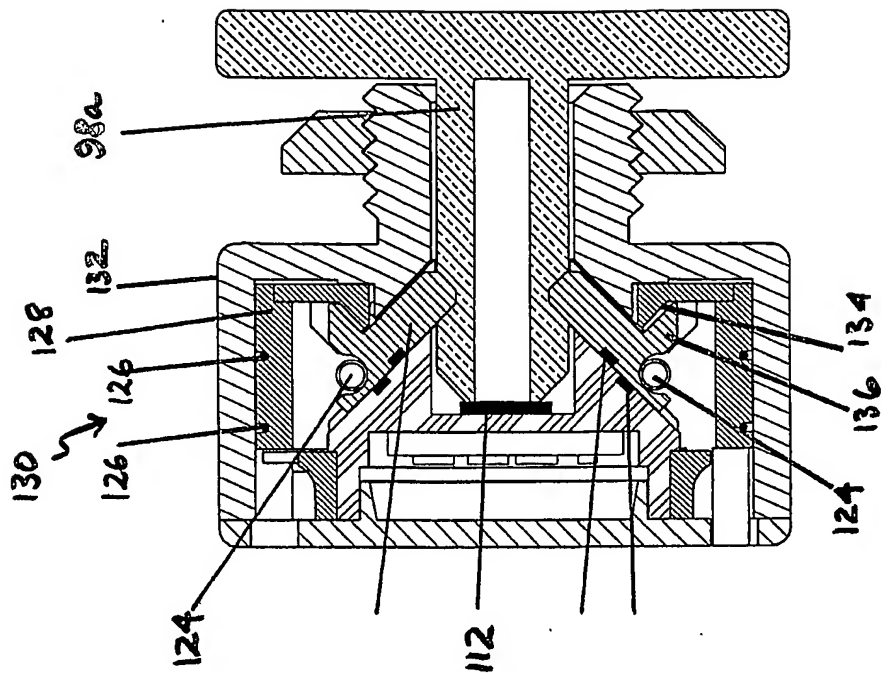


FIGURE 18

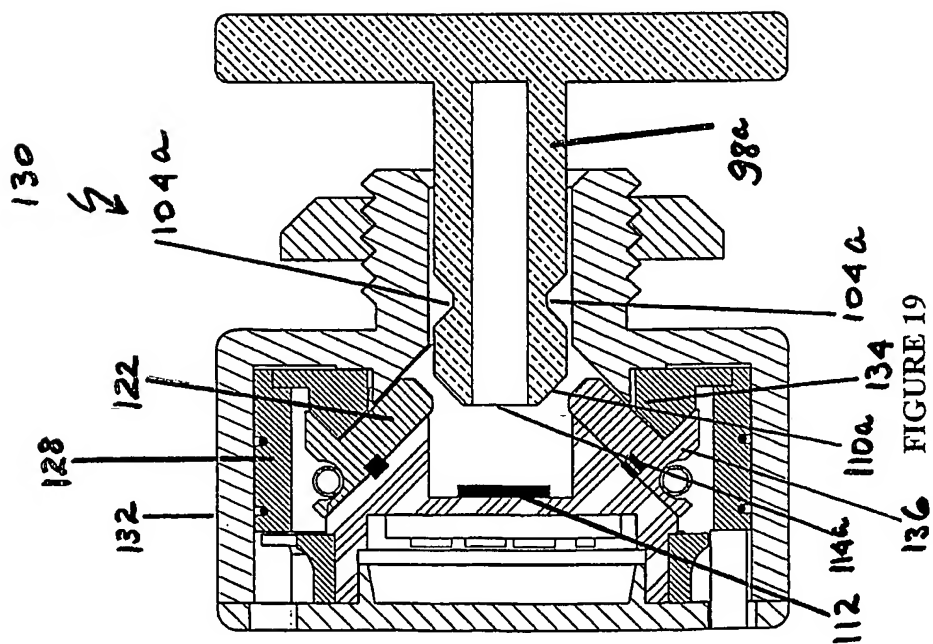
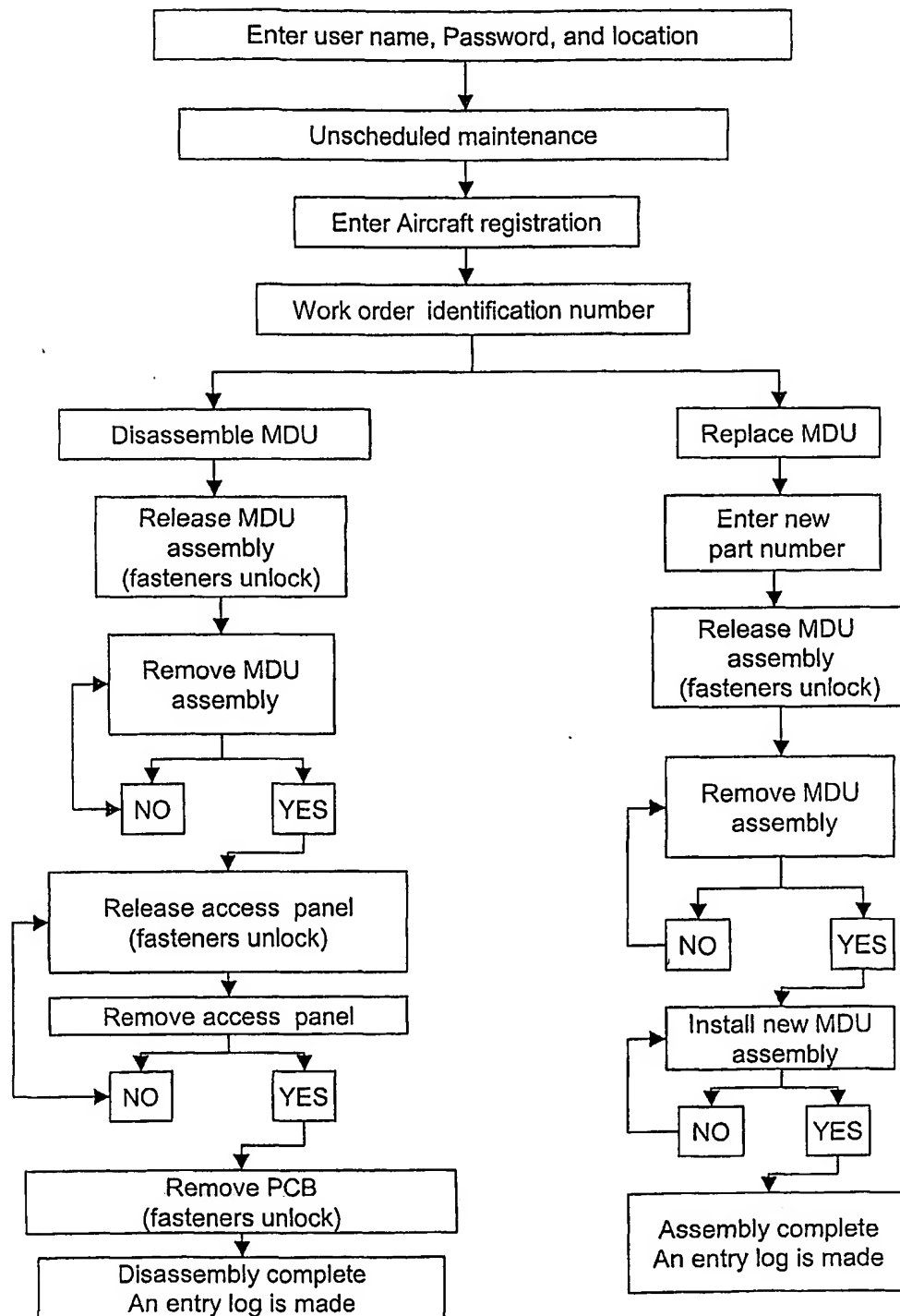


FIGURE 19

11/11

Instrumentation servicing procedure**FIGURE 20**

INTERNATIONAL SEARCH REPORT

International application No.

PCT/AU2004/000623

A. CLASSIFICATION OF SUBJECT MATTER

Int. Cl. ⁷: B23P 19/04, B23Q 16/00, B25B 27/00

According to International Patent Classification (IPC) or to both national classification and IPC

B. FIELDS SEARCHED

Minimum documentation searched (classification system followed by classification symbols)

Refer Electronic Database consulted below.

Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched

Electronic data base consulted during the international search (name of data base and, where practicable, search terms used)

Derwent World Patent Index: B23P-019, B23Q-016, B25B-027 and keywords : SENS+ AND (SPACIAL OR CONNECT+)

C. DOCUMENTS CONSIDERED TO BE RELEVANT

Category*	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
X	US 6141867 A (FUKADA et al.) 7 November 2000 See the whole document	1 - 24
X	US 6081993 A (RANKIN) 4 July 2000 See the whole document	1 - 24
X	US 5771578 A (KING, Jr. et al.) 30 June 1998 See the whole document	1 - 24
X	US 5531009 A (GIVLER) 2 July 1996 See the whole document	1 - 24

☒ Further documents are listed in the continuation of Box C

☒ See patent family annex

* Special categories of cited documents:

"A" document defining the general state of the art which is not considered to be of particular relevance

"E" earlier application or patent but published on or after the international filing date

"L" document which may throw doubts on priority claim(s) or which is cited to establish the publication date of another citation or other special reason (as specified)

"O" document referring to an oral disclosure, use, exhibition or other means

"P" document published prior to the international filing date but later than the priority date claimed

"T" later document published after the international filing date or priority date and not in conflict with the application but cited to understand the principle or theory underlying the invention

"X" document of particular relevance; the claimed invention cannot be considered novel or cannot be considered to involve an inventive step when the document is taken alone

"Y" document of particular relevance; the claimed invention cannot be considered to involve an inventive step when the document is combined with one or more other such documents, such combination being obvious to a person skilled in the art

"&" document member of the same patent family

Date of the actual completion of the international search
13 August 2004

Date of mailing of the international search report
19 AUG 2004

Name and mailing address of the ISA/AU
AUSTRALIAN PATENT OFFICE
PO BOX 200, WODEN ACT 2606, AUSTRALIA
E-mail address: pct@ipaustalia.gov.au
Facsimile No. (02) 6285 3929

Authorized officer

C. NGUYEN-KIM

Telephone No : (02) 6283 2121

INTERNATIONAL SEARCH REPORT

International application No.

PCT/AU2004/000623

C (Continuation). DOCUMENTS CONSIDERED TO BE RELEVANT		
Category*	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
X	US 5331831 A (SCHNEIDER) 26 July 1994 See the whole document	1 - 24
X	US 4731923 A (YAGI et al.) 22 March 1988 See the whole document	1 - 24
X	US 4675993 A (HARADA) 30 June 1987 See the whole document	1 - 24
X	US 4654964 A (SCHNEIDER et al.) 7 April 1987 See the whole document	1 - 24
X	US 4603329 A (BANGERTER et al.) 29 July 1986 See the whole document	1 - 24
X	US 4575931 A (JONCA) 18 March 1986 See the whole document	1 - 24
X	US 4453303 A (LEDDET) 12 June 1984 See the whole document	1 - 24
X	EP 1088626 A (SCHULTE-BUNERT GMBH) 4 April 2001 See the whole document	1 - 24
X	DE 19510856 A (PRIELOG LOGISTIK GMBH PRIEROS) 19 September 1996 See the whole document	1 - 24
X	Derwent Abstract no. 1999-554364/47, Class P56;V04, JP 11239925 A (MATSUSHITA DENKI SANGYO KK) 7 September 1999 See the English language version in US 6735856 B	1 - 24
X	Derwent Abstract no. 99-251075/21, Class P56, JP 11074693 A (TDK CORP) 16 March 1999 See the abstract	1 - 24

INTERNATIONAL SEARCH REPORT

Information on patent family members

International application No.

PCT/AU2004/000623

This Annex lists the known "A" publication level patent family members relating to the patent documents cited in the above-mentioned international search report. The Australian Patent Office is in no way liable for these particulars which are merely given for the purpose of information.

Patent Document Cited in Search Report		Patent Family Member			
US	6141867	EP	0926782	JP	11191329
US	6081993	JP	2000263349	US	6279223
US	5771578	CA	2214883	EP	0831556
				JP	10099927
US	5531009	US	5203855	US	5210935
		US	5216819	US	5213454
		US	5231747	US	5231754
		US	5259104	US	5263236
		US	5379508	US	5361484
		US	5404633	US	5407003
		US	5566747	US	5577315
		US	5621963	US	5621963
		US	5752306	US	5774968
US	5331831	AU	65204/94	CA	2156261
		WO	9421421	EP	0689494
US	4731923	CN	87101876	DE	3708119
		GB	2189719	FR	2603766
		JP	63161635	JP	62229900
				JP	63161700
US	4675993	CA	1240403	DE	3546216
		GB	2169422	FR	2575357
				JP	61152100
US	4654964	EP	0187327	JP	61163694
US	4603329	DE	3413474	JP	60082990
US	4575931	EP	0123574	FR	2542934
				JP	60000083
US	4453303	EP	0072277	ES	8305616
		JP	58036779	FR	2510933
EP	1088626	DE	19946401		
DE	19510856	DE	19604028		
JP	11239925	US	6735856		
JP	11074693				
Due to data integration issues this family listing may not include 10 digit Australian applications filed since May 2001.					
END OF ANNEX					